



Stress Measurements in Bulk Metallic Glass Matrix Composites

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Demonstration of the elastic limit of BMG







- Vitreloy1 (Vit1) Developed at Caltech in early 90's
 - Composition: Zr_{41.2} Ti_{13.8} Cu_{12.5} Ni₁₀ Be_{22.5}
 - Critical cooling rate about 1°C per second
 - Glass transition temperature: 350°C
 - Maximum size approaching 100 mm in thickness
- Mechanical properties
 - Yield strength: 1.9 GPa, Young's modulus: 96 GPa
 - Elastic strain limit: 2%
 - No ductility. Due to formation of macroscopic shear bands
 - Fracture toughness: up to 55 MPa m¹/₂ (similar to maraging steel)

Why BMG matrix tungsten fiber composites?





Strain, %

- BMG/Tungsten fiber composites
 - Same ultimate stress as monolithic Vit1
 - Large increase in ductility
 - Knee in stress strain curve: Yielding in tungsten?

Neutron diffraction equipment



- Spectrometer for MAterials Research at Temperature and Stress (SMARTS)
- Schematic set-up for *in-situ* compression loading
- Measurement time is about 10-20 minutes per load level
- Measure elastic strains in two directions simultaneously
- Bulk measurement contrary to conventional X-ray measurements

Neutron diffraction measurements





- Diffraction patterns from BMG tungsten fiber composite sample (80%)
- Highly textured fibers; hh0 texture for wire drawn bcc metals
- Good statistics from short count times
 - $R_{wp} \approx 6-8\%$, strain error bar $\approx 15 \ \mu\epsilon$

Finite element model: inputs



• BMG

- Young's modulus: 96 GPa
- Poisson's ratio: 0.36
- Yield stress (Mohr-Coulomb): $\tau_c = 946 - 0.04\sigma_n$ [MPa]
- No hardening
- Tungsten
 - Young's modulus: 410 GPa
 - Poisson's ratio: 0.28
 - Yield stress (Von Mises): $\sigma_v = 1305$ [MPa]
 - Hardening deduced from the neutron data

R. D. Conner, R. B. Dandliker and W. L. Johnson, Acta Mater., vol. 46(17), pp. 6089-6102, 1998
Lewandowski J. J. and Lowhaphandu P., Phil. Mag. A., in print
A. Saigal and G.G. Leisk, Mat. Sci & Eng. A, vol. 237, pp. 65-71, 1997
Y. He, R. B. Schwarz and D. G. Mandrus, J. Mater. Res., vol. 11, p. 1836, 1996







- Measured and calculated thermal residual stresses
 - No plasticity observed in either phase during the cooling cycle
 - Good agreement for the tungsten fibers
- Determine the thermal residual stresses in both phases from the validated FEM calculations

In situ loading measurements

CHNOI



 Good agreement with measured data ⇒ Use the FEM to obtain information not measurable by neutron diffraction

Finite element model: Additional insights





- Von Mises stresses at highest load level (1000MPa for 20%, 1600MPa for 80%)
 - Stress concentration for the 80% precursor for shear band formation
 - No appreciable variation of Von Mises stress in fibers (not shown)

Finite element model: Macroscopic yielding





- Calculated and measured macroscopic yielding
 - Capture the increase in yield point (knee) with increasing fiber volume fraction
 - Does not capture the perfect plastic behavior at high strains

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- Combination of diffraction measurements and modeling (e.g. FEM) allows for the determination of internal and residual stresses
 - Even for the BMG, whereas neutron diffraction alone does does not provide any data for the glass
- Load sharing during uniaxial compression
 - Knee in macroscopic loading curves: Decrease in elastic strain development in the tungsten fibers clearly shows that the fibers are yielding

Future Work

- The ductility of the composites are linked to formation of multiple shear bands caused by the obstruction of the shear bands by the tungsten fibers
 - Planned studies of formation of multiple shear bands *in situ* in composites using synchrotron x-ray tomography



BMG matrix tungsten fiber composites





- Measured volume fractions deviates slightly compared to nominal volume fractions
- "Agglomeration" seen for all volume fractions
- "Stacking faults" seen for the 80% sample



 $\lambda = 2 d \sin \theta$

- Fixed λ ; Reactor (steady state). Measure intensity as function of angle
- Fixed θ : TOF (spallation). Measure intensity as function of time-of-flight



• Differences in lattice spacing => Only Elastic Lattice Strain of Crystalline Phase

$$\varepsilon_{hkl}^{el} = \frac{d_{hkl} - d_{hkl}^{0}}{d_{hkl}^{0}} = \frac{d_{hkl}}{d_{hkl}^{0}} - 1$$

Finite element model input



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Finite element model



- Full 3D model due to loading along fibers
 - Plane strain by keeping planes perpendicular to fibers plane
 - Unit cell model
 - Brick 2nd order elements
- Hexagonal stacking in all models to accommodate high volume fractions
- Thermal cooling cycle
 - Choose ∆T (335°C) to give good comparison with measured thermal residual stresses
 - Same ΔT for all volume fractions
 - Close to T_g (350°C)

